On-orbit Results of Pointing, Acquisition, and Tracking for the TBIRD CubeSat Mission

Kat Riesing¹, C. Schieler¹, J. Chang¹, N. Gilbert¹, A. Horvath¹, R. Reeve¹, B. Robinson¹, J. Wang¹, P. Agrawal², R. Goodloe²

¹MIT Lincoln Laboratory ²Terran Orbital

SPIE Free-Space Laser Communications XXXV 30 January 2023



DISTRIBUTION STATEMENT A. Approved for public release. Distribution is unlimited. This material is based upon work supported by the National Aeronautics and Space Administration under Air Force Contract No. FA8702-15-D-0001. Any opinions, findings, conclusions or recommendations expressed in this material are those of the author(s) and do not necessarily reflect the views of the National Aeronautics and Space Administration.

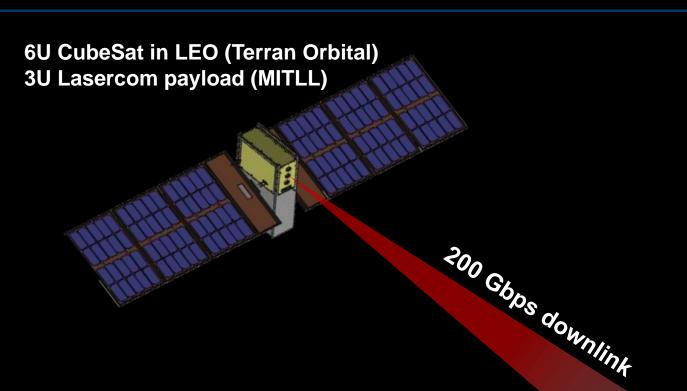
© 2022 Massachusetts Institute of Technology.

Delivered to the U.S. Government with Unlimited Rights, as defined in DFARS Part 252.227-7013 or 7014 (Feb 2014). Notwithstanding any copyright notice, U.S. Government rights in this work are defined by DFARS 252.227-7013 or DFARS 252.227-7014 as detailed above. Use of this work other than as specifically authorized by the U.S. Government may violate any copyrights that exist in this work.



Terabyte Infrared Delivery (TBIRD)



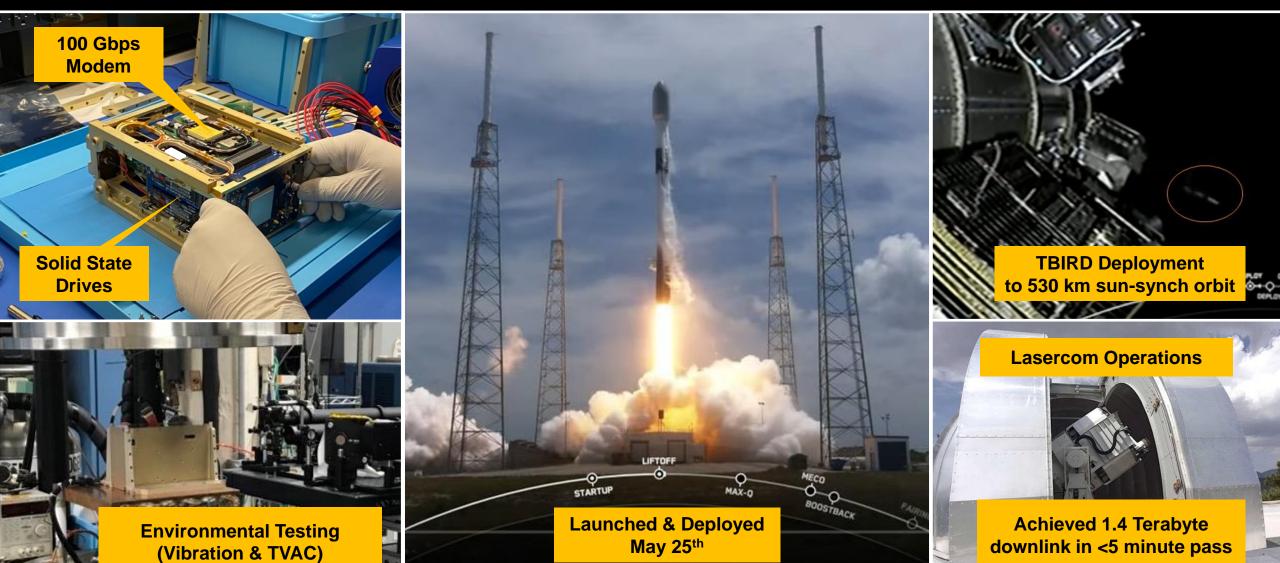


- Leverage fiber telecom equipment for 200 Gbps burst delivery (TBs per pass)
- Demonstrate robust data transfer through atmospheric channel
- 3U lasercom terminal payload hosted on 6U CubeSat
 - NASA Small Sat Pathfinder Tech Demo

Ground terminal at OCTL in Southern California (MITLL & JPL)

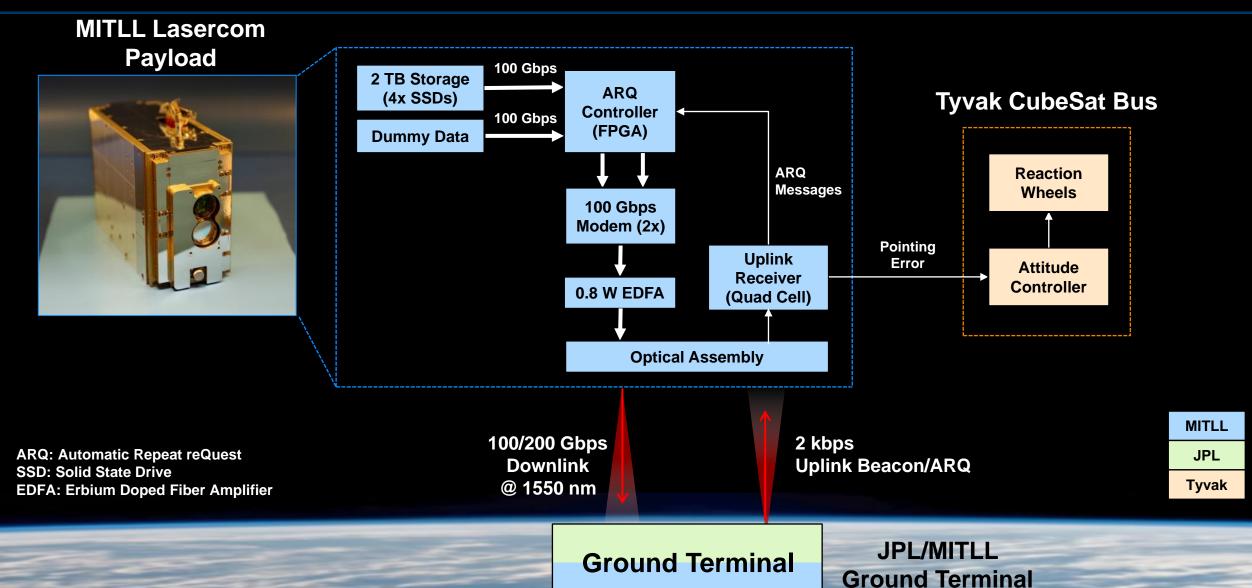


TBIRD Demonstration





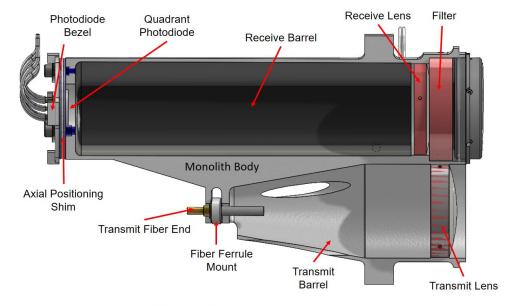
Communication and Body-Pointing Architecture

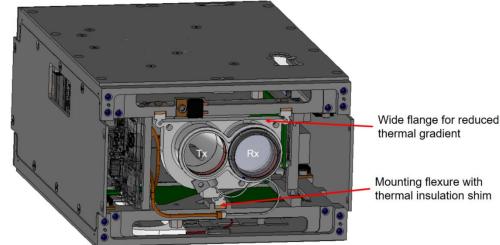




Optical Assembly

- Bistatic design to reduce complexity
- Payload draws 105 W in 200 Gbps mode, resulting in self-heating
- Custom optical assembly was designed to maintain Tx/Rx alignment stability
 - Thermal isolation from electronics
 - Large thermal mass
- Quad sensor enables spatial tracking of uplink
 - Offset tracking to compensate for point ahead and Tx/Rx misalignment







State-of-the-Art in Nanosatellite Pointing

- Nanosatellite pointing has improved 1000x in the last decade: ~2° to ~0.002°
 - COTS options available from individual components to full ADCS to entire bus
 - Still relatively few precision pointing CubeSat missions
- ASTERIA mission is best pointing to date (inertially pointed)
- OCSD demonstrated precision ground tracking for lasercom from a CubeSat (beaconless)
- TBIRD is the first lasercom mission to demonstrate architecture of body-pointing with closedloop payload feedback

Year	Mission	Organization	ADCS Vendor	Pointing Accuracy
2013	BRITE	UTIAS-SFL	Sinclair Interplanetary	262 urad (RMS, 2-axis)
2016	MinXSS	UC Boulder	Blue Canyon Technologies	73-204 urad (3σ, 1-axis, axis dependent)
2017	ASTERIA	JPL	Blue Canyon Technologies	8.7-22 urad (RMS, 1-axis, axis dependent)
2018	AeroCube-7/OCSD	Aerospace Corporation	In-house	419 urad (3σ, 1-axis)
2022	TBIRD/PTD-3	MITLL/Terran Orbital	Terran Orbital	20-35 urad (RMS, 1-axis, axis dependent)



Pass ConOps



Acquisition

- Bus open-loop points within field of view of TBIRD quad sensor
- Ground station illuminates TBIRD by openloop pointing from ephemeris (TLE)
- Bus can perform scan as a contingency

Sensor FOV ±5 mrad

Uplink beam 600 urad FWHM

FOV: Field of view

FWHM: Full width half max

OCTL: Optical Communications Telescope Laboratory

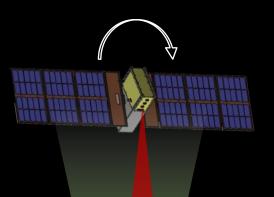
TLE: Two Line Element

JPL/OCTL
Ground terminal



Pass ConOps





Tracking

- Body-pointing only, no additional actuators
- TBIRD sends 2-axis pointing feedback to bus
- Bus tracks payload feedback to support high-rate downlink

Downlink beam 380 urad FWHM



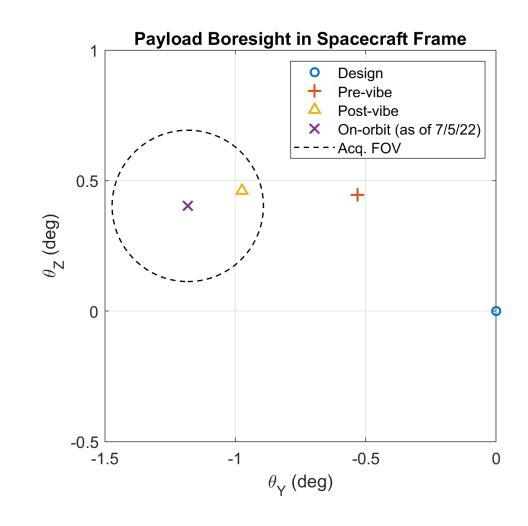
FWHM: Full width half max **OCTL:** Optical Communications Telescope Laboratory

JPL/OCTL
Ground terminal



Initial Uplink Detection

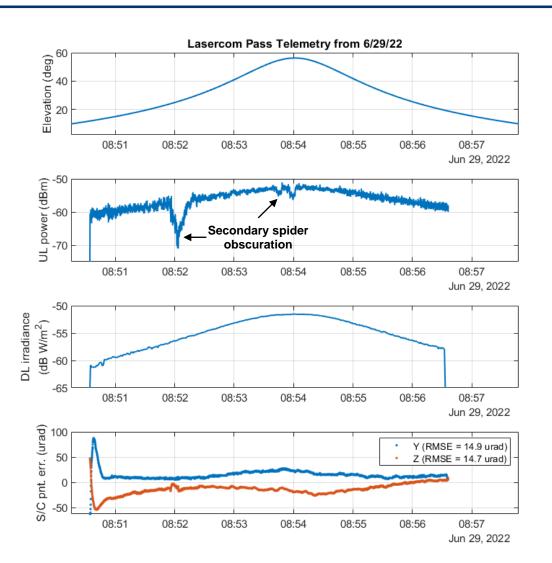
- Required steps for uplink detection
 - 1. Ground station illuminates spacecraft
 - 2. Spacecraft points at ground station within sensor field of view
- Primary sources of pointing error
 - Ground station: ephemeris error
 - Spacecraft: payload boresight uncertainty
- Payload boresight measured on ground, but some shift expected due to launch
- Payload boresight shift was within sensor FOV enabling acquisition without scan





Sample Telemetry from 6/29/22, 500 GB Downlinked

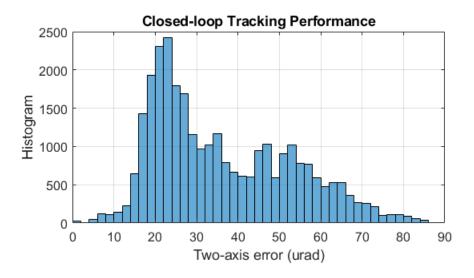
- Uplink detection occurs at 12° elevation (~1600 km), pass ends at 20° descending
- Bus takes about 20 seconds to pull-in
- Observed uplink power dips are due to telescope spider interference
- In this pass, pointing error is about 15 urad RMS per axis
- Pointing error dominated by a low frequency (<0.02 Hz) drift

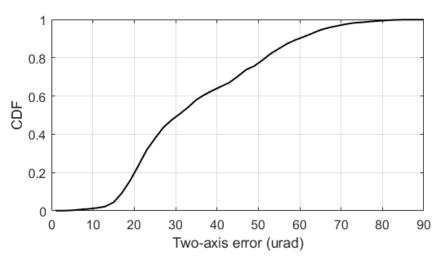




Closed-loop Tracking Performance

- Data from 18 passes, ~50 minutes of data at 10 Hz
 - Excludes off-nominal passes
 - Excludes pull-in time
- Two-axis pointing statistics:
 - 50% of time error <30 urad
 - 90% of time error <60 urad
- Per-axis pointing statistics:
 - Y RMS error 35 urad
 - Z RMS error 20 urad
- Low-frequency pointing error could be reduced with controller modification
- Pointing loss typically <0.1 dB for 380 urad FWHM beam



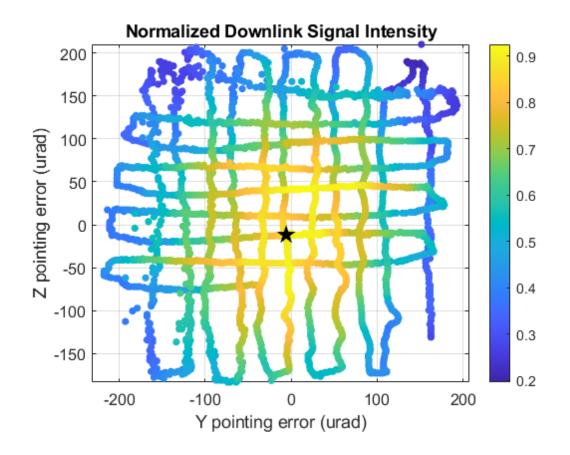




Downlink Beam Characterization

- Two passes were used to perform downlink beam characterization
 - Offsets injected into payload feedback to steer the bus in a grid
 - Power measured at ground station
- Measured beamwidth of 380 urad FWHM
- Confirmed Tx/Rx alignment has been maintained to within 20 urad







Summary

- TBIRD/PTD-3 launched in May 2022 and has operated successfully for six months
- Key achievements:
 - 100/200 Gbps downlinks from LEO
 - Downlinked >1 TB error-free in a pass
 - Demonstrated 20-35 urad (RMS, per axis) closed-loop body pointing on a Cubesat
- Mission will continue in 2023

